CHAPTERII

THEOREFICAL CONSIDERATIONS

PART 1 Dynamomoter Requirements

Two requirements that are always in exposition in dynamometer design are sensitivity and rigidity. The sensitivity of a good research dynamometer chould be such that determinations are accurate to within * 1%. That is, if a dynamometer is designed for a mean force of 100 parada, one pound increments should be easily readable.

Some deformation is accordated with the operation of every dynamometer. However, a dynamometer should be rigid enough so that the cutting operation is not influenced by the accompanying deflections. Proquently, the deminating stiffness criterion in the <u>natural frequency</u> of the dynamometer. All machine tools operate with some vibrations, and in certain outting operations these vibrations may have large emplitudes.

In order that the recorded force (or the actual out) shall not be influenced by any vibrating notion of the dynamometer, it's natural frequency must be large (at least 4 times as large) compared to the frequency of the exciting vibration. For purposes of analysis any dynamometer can be reduced to a mass supported by a spring. The natural frequency (8, of such a system is equal to

$$v_{\mathbf{R}} = \frac{1}{2\pi} \left(\frac{\mathbf{E}}{\mathbf{N}}\right)^{\mathbf{F}_{\mathbf{A}}} \mathbf{c}_{\mathbf{p}} \mathbf{p}_{\mathbf{n}}$$

where K is the spring constant in 1b/in

In torne of the supported weight of the dynamoter (w)

In general, a dynamical must measure at loast two force components in order to determine a two dimensional resultant cutting force. In a three dimensional cutting operation, three force components are necessary, while in drilling or topping, only a torque and a thrust are required. It is usually most convenient to measure force relative to a set of rectangular coordinates (x,y,z) and it is advisable that there be no cross-conditivity between these components. That is an applied force in the z direction should give no reading in y or z directions. If antual interference of the force measuring classests exists, determination of the force economic requires the solution of simultaneous equations.

This provents the immediate interpretation of the date, when certain of the electric transducers are smitchly located and connected, unwanted strain component can often be concelled electrically. How this may be done will be subnequently illustrated.

It is convenient to use a system having a linear <u>enlibration</u>.

In such a case, the force is determined with the precision with which a strain increment can be measured relative to an arbitrary datum.

If the system is not linear, it is then necessary that the zero load point be accurately known as well as the strain increment. This introduces an additional quantity that must be carefully peasured.

A dynamometer should be stable with respect to time, temperature, and butidity once a calibration is made & it should only have to be checked occasionally. Henry existing dynamometers use devices for the separation of force components which involve friction, (i.e. relieve, balla and midding surfaces) As friction condition are usually variable due to dirt, temperature, etc. such instruments are of limited usefulness. (2)

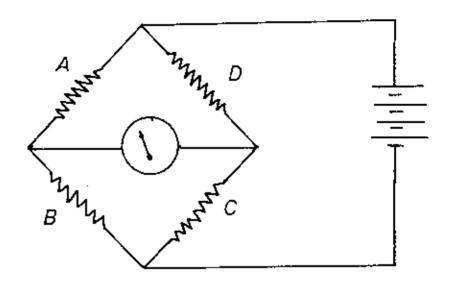


FIG. 2.1 Wheatstone bridge

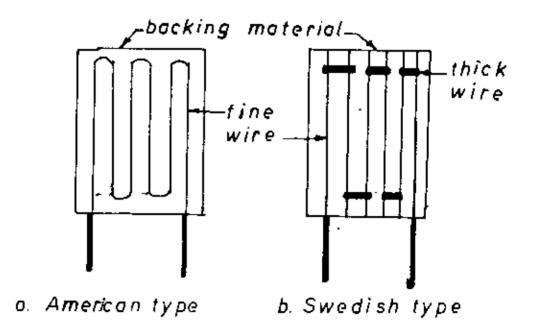


FIG. 2.2 STRAIN GAUGE

<u>Force measurement</u> a variety of devices have been used to measure forces in dynamometers but the two component lathe dynamometers as designed here is a wire resistance strain gauge type.

Principle of operation of strain gauge A strain gauge works on the principle discovered by Lord Kelvin that when a conductor is stretched, its resistance increases. If, there fore, a conductor (the strain gauge) is bonded intimately to the surface of a specimen, then any subsequent strain in the specimen is experienced by the gauge, resulting in a change of resistance.

As the strains normally experienced in structures are small (mirco inches / inch) The change in resistance of a strain gauge is also small (micro chas / ohm) and therefore a Wheatstone Bridge net-work is usually used to measure these pery small values Fig. 2.1.

The four arms of the bridge may be made up in various ways according to the type of measurement required and the instrumentation used. For measurement of strain at a point, a single atrain gauge is used as one are A., with three fixed resistances making up the bridge B., C. and D. Host static strain recording instruments have two resistors built in C. and D. and the other two arms are made up of the active gauge cemented to the specimen and a dummy gauge (compensating gauge) comented to a piece of similar material which experiences the same temperature conditions as the test specimen but is unstrained. This is called a half bridge. It gives temperature compensation and climinates apparent strain due to thermal expansion or contraction of specimen. Instruments incorporating two fixed resistors are usually of the null balance typs, in which out of balance eignals are corrected by means of a slide wire which is calibrated directly in percentage change. This system

has the adventage of being independent of in-put voltage, but it is only outtable for static or alouly changing strains. For measuring loods or displacements in a calibrated system, the signal con be increased by making more than one are of the bridge active. In this case otrain gauges are used for all four area of the bridge, known as full bridge, and the mignal is fed directly to a galvenometer. a galvanamater recorder, or a sensitive micro-ammeter, The options for moneuring lands or deflections is the cantilever bear whore gauges A. and C. are mounted on the upper surface and C. and D. on the lower surface make all four arms active. The change of resistance of gauges A. and C. is equal and opposite to that of gauge B. and B. so that the cignal obtained is four times that from a single active gauge and is usually sufficiently large to drive a low realstance 0-50 micro-armoter direct with an energiaing voltage of between 6 and 12volto DC. Alternatively, the signal can be put directly into a galvanguetor recordor to give a personent trace at frequencies up to 3,000 cyclos/sec.

The disadvantage of the above systems, which are called <u>deflection</u> methods, as opposed to the <u>null belance</u> method, is that the signal obtained is directly proportional to the applied voltage, saking some kind of voltage stabilized supply necessary for occurate work. (4)

Bonded strain gauge construction Bonded strain gauges consist of fine metal wire of 0.0004 to 0.007 in. thickness, depending on the type of gauge, forming a grid pattern which is mounted on a backing material of shown in fig. 2.2

The gauges are communicatived with differing metal and backing unterials as necessary.

Gauge factor The gauge factor is the relationship between the change in resistance of the gauge and the strain, and is defined as

transmitting strain to the dumny block. Often it is preferable to maintain the dumny block of constant temperature and calibrate the active gauge against temperature.

If the two coldered joints on a gauge are at the same temperature then the thornal o.a.f's will be equal and opposite and will cancel out.

When working at elevated temperatures a chack for stroy e.m.f's chould always be made by disconnecting the power surply to the bridge and varying the temperature of the specimen. If there are sero-shifts with temperature and those can not be eliminated, a graph of soro-shift against temperature must be sade and used to correct readings node under load.

Copper has a high temperature confficient so that, although the lead resistances are usually low, the errors introduced by leads in the bridge circuit being either in an elevated temperature area or cubjected to varying temperatures can be expreciable,

It is always proforable to complete a bridge at the point of measurement in order to keep the lengths of wire in the bridge as short as possible. The errors introduced by long leads ex-ternal to the bridge circuit are generally negligible a part from voltage drop.

Local in high temperature zones should always be kept as short as messible. They should be establed for length and resistance, and bound, together so that they all pass through the case environment. (4)

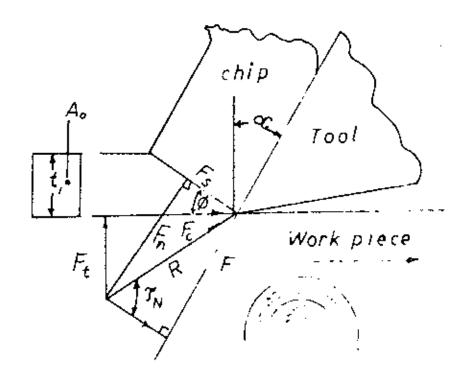


FIG. 2.3 Force diagram for orthogonal cutting

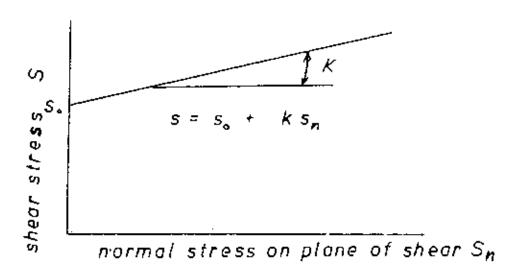


FIG. 2.4 Dependance of son s_nassumed in Merchants second theory

PART 2 Nechanice of cutting

Theory of Ernst and Merchant

Although an attempt had been made to solve this problem by Piispanen in 1937 the first complete analysis resulting in a so-called shear angle solution was presented by Ernst and Merchant.

(Fig. 2.3 . p. 11)

R = resultant tool force

F s cutting force

F, = thrust force

F shoar force on shear plane

F_ = normal force on chear plane

r = friction force on rake face

ø = ehear angle

T s mean friction angle

A = cross-section area of uncut chip

t, = undeformed chip thickness

t₂ = chip thickness

In this analysis the clip is assumed to be a rigid body held in equilibrium by the action of the forces transmitted across the chiptool interface and across the shear plane.

For convenience, in Fig. 2.3 the resultant tool force R. is shown acting at the tool cutting edge and is resolved into components R. and F. in directions normal to and along the tool face respectively and into component F_n and F_n normal to and along the chear plane respectively. The cutting (F_c) and thrust (F_t) components of the resultant tool force are also shown.

It is assumed that the whole of the resultant cutting force is transmitted across the chip-tool interface and that no force acts on the tool edge or flank (i.e. the ploughing force P=0)

The basis Ernot and Merdhant's theory was the suggestion that the shear angle β would take up such a value as to make the work done in cutting a minimum. Since, for given cutting conditions, the work done in cutting is proportional to F_c it was necessary to develop an expression for F_c in terms of β and then to obtain the value of β for which F_c is a minimum.

From Fig. 2.3

where S = nhear strength of material on the chear plane $A_{g} = area$ of shear plane

Stress S = S This assumption agreed with the work of Bridgman where, in experiments on poly crys-talline metals, the shear strength was shown to be dependent on the normal stress on the plane of shear

Now from Fig. 2.3

$$T_n = R \sin (\phi + T - \alpha)$$
(8)

and

$$\mathbf{f}_{\mathbf{n}} = \mathbf{S}_{\mathbf{n}} \mathbf{A}_{\mathbf{0}} \\
= \mathbf{S}_{\mathbf{n}} \mathbf{A}_{\mathbf{0}} \\
= \mathbf{S}_{\mathbf{n}} \mathbf{A}_{\mathbf{0}} \\
= \mathbf{S}_{\mathbf{n}} \mathbf{A}_{\mathbf{0}}$$

From (8) and (9)

$$\delta_n = \frac{\sin \theta}{A_0} R \sin (\theta + T - \infty)$$
(10)

Combining equation (3) and (10)

and from equations (7) and (11)

$$S = \frac{S_0}{1-K \tan (\emptyset + T - \omega)}$$
 (12)

This equation shows how the value of S may be affected by changes in β and is now inserted in equation (5) to give a new equation for F, in terms of β ;

It is now assumed that K and S are constants for the particular work material; and that A and are constants for the cutting operation. Thus eq. (15) may be differentiated to give the new value of

Ap = cross-sectional area of undeformed chip

T = mean angle of friction between this and tool (=ten T/T)

From (1) and (2)

$$H = \frac{8 \text{ A}_0}{\sin \theta} \frac{1}{\cos(\theta + f - \infty)}$$

Now by geometry

$$F_{c} = R \cos(\mathcal{T} + \kappa) \qquad (4)$$

Hence from (5) and (4)

$$F_{c} = \frac{S A_{0} \cos(\tau - \alpha)}{\sin \beta \cos(\beta + \tau - \alpha)}$$

Equation (5) may now be differentiated with respect to \emptyset and equated to zero to find the value of \emptyset for which F_c is a minimum. The required value is given by

$$2 \cancel{b} + 7 - \infty = 90^{\circ} \qquad (6)$$

Merchant found that this theory gave good agreement with experimental results obtained when cutting cynthetic plastics but gave poor agreement for steel machined with a sintered carbide fool.

It should be noted that, in differentiating equation (5) with respect to \$\mathbb{E}\$, it was assumed that \$A_0\$, and would be in-dependent of \$\mathbb{J}\$. On reconsidering these assumptions, Merchant decided to include in a new theory the relationship

which indicates that the shear strength of the material increases linearly with increase in normal stress on the shear plane (Tig.2.3) at zero normal.

The resulting expression is

where $C = \cot^{-1} K$ and is a constant for the work material.